# Multivariable Parametric Cost Model for Ground and Space Telescope Assemblies H. Philip Stahl

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#### Motivation

Decadal 2020 is soliciting White Papers which require cost estimates. Some of those white papers may use one or another of my published parametric cost models (see Bibliography) to provide a basis of estimate for a total project cost or to compare cost between major architectural cost drivers or to justify technology development investment. Given that these models have evolved over the course of my publication history (i.e. some are more right than others) and that recently a multivariable parametric model has been developed that explains the cost of both ground and space telescopes, it was deemed important to summarize this new model in a White Paper to the Decadal 2020. And, to make explicit what this model can and cannot do.

## **Summary**

The NASA Marshall Space Flight Center has developed a 5-parameter cost model that explains 92% (Adjusted R2) of the cost variation in a database of 47 total ground and space telescope assemblies (OTA), where OTA is defined to include primary & secondary mirrors, and structure.

OTA\$ (FY17) = \$20M x 30 x D x  $\lambda$  x T (-0.5) (-0.25) x (-0.028) (Y-1960)

where:

(S/G) = 1 for Space OTAs

= 0 for Ground OTAs

D = Effective Telescope Aperture Diameter

 $\lambda$  = Wavelength of Diffraction Limited Performance

T = Operating Temperature

YOD = Year of Development, i.e. start of Phase C/D or Issue of Contracts.

Some high level implications of the model are:

- Space telescopes are approximately 30X more expensive than ground telescopes with the same aperture diameter and diffraction limited performance.
- Larger telescopes are more expensive than smaller telescopes.
- UV diffraction limited telescopes are more expensive than IR telescopes.
- Cryogenic telescopes are more expensive than warm telescopes.
- And, telescope cost has historically decreased by approximately 50% every 25 years presumably due to technology advances.

Note: The MSFC OTA Cost Model only provides an estimate of the most likely or 50% probably cost. The uncertainty on this estimate is nearly 50%. Thus, to be conservative, use the 84% probably cost – multiply the 50% estimate from the model by 1.5.

Additionally, the data indicates that telescope mass is not a good indication of telescope. And, the data is inconclusive regarding whether segmentation increases or decreases telescope cost.

#### **Disclaimers**

Parametric cost models cannot predict the cost of a future system. They are backward looking. They describe how actual historical costs vary as a function of quantifiable technical and programmatic parameters. The only forward prediction that a cost model can make is to provide guidance as to how the cost of a potential future system might scale relative to an existing historical system.

Parametric cost models do not predict the cost of a specific mission or any component of that mission. They provide an estimate of the most probable cost and an estimate of the uncertainty of that cost. One

reason for estimation uncertainty is because non-technical issues such as managerial decisions and funding profiles can have profound impacts on cost.

A parametric cost model is only as good as its database. Inconsistencies and inaccuracies in a cost model are the results of insufficient data quantity or diversity, inconsistencies in definitions, or data errors or inaccuracies. In our own research, every time the database was changed, the cost model changes slightly. For example, since the AURA "Space Astrophysics Landscape for the 2020s and Beyond" meeting, we added the CALIPSO telescope to the database. This one change resulted in small increases to all the exponents in the 3<sup>rd</sup> decimal place and reduced all to p-values by a factor of 2. For this reasons, no one should ever use or accept the output of a cost estimating tool without understanding the tool's underlying database. Thus, the fundamental challenge of cost modeling is developing a parametric model that includes the most important parameters. And, to do this requires a database with sufficient samples and data diversity to yield statistically meaningful results, and engineering judgment to interpret the results.

While there is a tendency to focus on the OTA cost, in reality OTA cost is a small portion of any given mission – only about 10 to 15%. Instruments and spacecraft are each typically a larger portion of the total mission cost than the telescope.

### **Definitions**

The MSFC multivariable model estimates the most likely cost for ONLY an Optical Telescope Assembly (OTA). Where an OTA is defined as the mission subsystem that collects electromagnetic radiation and focuses it (focal) or concentrates it (afocal) into the science instruments. An OTA consists of the primary mirror, secondary mirror, auxiliary optics and support structure (such as optical bench or truss structure, primary support structure, secondary support structure or spiders, straylight baffles, mechanisms for adjusting the optical components, electronics or power systems for operating these mechanisms, etc.). An OTA does not include science instruments or spacecraft.

Using NASA terminology, the model estimates ONLY Contractor Cost for Phase A to D, i.e. design, development, integration and test. The model cost estimate does not include Pre-Phase A (i.e. formulation) costs, Phase E (launch/operation) costs, government labor costs, government furnished equipment (GFE) or existing contractor infrastructure or other non-contracted costs.

The model ONLY estimates 'first-unit' cost.

A cost model is a statistical correlation between an item's historical cost (dependent variable) and quantifiable technical or programmatic parameters (independent variables). Statistically significant correlations are called Cost Estimating Relationships (CERs).

# **Database**

The MSFC multivariable parametric telescope cost model is based on 47 telescopes (27 space and 20 ground) out of a total database of 72 telescopes (46 space and 26 ground).

The MSFC Space OTA database contains information on over 47 different cost, programmatic and engineering parameters (Table 1) for 51 imaging and non-imaging space missions ranging from X-ray to UVOIR to FarIR (Table 2). The non-imaging missions include spectroscopic, LIDAR or radio/microwave systems.

Table 1: Space Telescope Database Cost, Programmatic and Engineering Parameters

| Primary Mirror Specific Information |             |  |  |  |  |  |  |
|-------------------------------------|-------------|--|--|--|--|--|--|
| PM Cost                             | \$ FY M     |  |  |  |  |  |  |
| PM Aperture Diameter                | meters      |  |  |  |  |  |  |
| PM Thickness                        | cm          |  |  |  |  |  |  |
| PM Surface Figure Error             | rms nm      |  |  |  |  |  |  |
| PM Material                         |             |  |  |  |  |  |  |
| PM Focal Length                     | meters      |  |  |  |  |  |  |
| PM F/#                              |             |  |  |  |  |  |  |
| PM Number of Segments               | #           |  |  |  |  |  |  |
| PM Segment Size                     | meter       |  |  |  |  |  |  |
| PM Mass                             | kg          |  |  |  |  |  |  |
| PM First Mode Frequency             | Hz          |  |  |  |  |  |  |
| Optical Telescope Assembly          | Information |  |  |  |  |  |  |
| OTA Cost                            | \$ FY M     |  |  |  |  |  |  |
| Diffraction Limit                   | micrometers |  |  |  |  |  |  |
| Transmitted WFE                     | nm rms      |  |  |  |  |  |  |
| OTA Structure First Mode            | Hz          |  |  |  |  |  |  |
| OTA Mass                            | kg          |  |  |  |  |  |  |
| System Focal Length                 | meters      |  |  |  |  |  |  |
| System F/#                          |             |  |  |  |  |  |  |
| FOV                                 | degrees     |  |  |  |  |  |  |
| Spatial Resolution                  | arc-seconds |  |  |  |  |  |  |
| Year of Development                 |             |  |  |  |  |  |  |
| Development Period                  | months      |  |  |  |  |  |  |
| Design Life                         | months      |  |  |  |  |  |  |
| TRL                                 |             |  |  |  |  |  |  |

| Total System Information  |             |  |  |  |  |  |  |
|---------------------------|-------------|--|--|--|--|--|--|
| Total Cost                | \$ FY M     |  |  |  |  |  |  |
| OTA + Thermal Cost        | \$ FY M     |  |  |  |  |  |  |
| Instrument Cost           | \$ FY M     |  |  |  |  |  |  |
| Operating Temperature     | K           |  |  |  |  |  |  |
| Total Mass                | kg          |  |  |  |  |  |  |
| OTA + Thermal Mass        | kg          |  |  |  |  |  |  |
| Instrument Mass           | kg          |  |  |  |  |  |  |
| Spectral Range Minimum    | micrometers |  |  |  |  |  |  |
| Spectral Range Maximum    | micrometers |  |  |  |  |  |  |
| Total Avg Input Power     | Watt        |  |  |  |  |  |  |
| Instrument Avg Power      | Watt        |  |  |  |  |  |  |
| Data Rate                 | Kbps        |  |  |  |  |  |  |
| Start Date                |             |  |  |  |  |  |  |
| Date of Launch            |             |  |  |  |  |  |  |
| Orbit                     | km          |  |  |  |  |  |  |
| Launch Vehicle            |             |  |  |  |  |  |  |
| Pointing Knowledge        | arc-second  |  |  |  |  |  |  |
| Pointing Accuracy         | arc-second  |  |  |  |  |  |  |
| Pointing Stability/Jitter | arc-sec/sec |  |  |  |  |  |  |
| # of Primary Mirrors      |             |  |  |  |  |  |  |
| # of Instruments          |             |  |  |  |  |  |  |
| # of Curved Optics        |             |  |  |  |  |  |  |
| Coating                   |             |  |  |  |  |  |  |

Table 2: Missions in Space Telescope Database

| Imaging      | Non-Imaging    | Not in Regression  | <u>Attached</u> |
|--------------|----------------|--------------------|-----------------|
| AFTA         | ACTS           | CCOR               | SOFIA           |
| COM_0.7      | CALIPSO/CALIOP | Commercial SiC .35 | HUT             |
| COM_1.1      | Cloudsat       | Commercial SiC .5  | UIT             |
| Herschel     | GALEX          | EO-1/ALI           | WUPPE           |
| HST          | ICESat/GLAS    | FUSE               |                 |
| IRAS         | IUE            | Imaging EUV        | <u>X-Ray</u>    |
| JWST         | MO / MOLA      | ISO                | EUVE            |
| Kepler       | OAO-B / GEP    | LandSAT-7          | Chandra         |
| MO / MOC     | SWAS           | SDO / AIA          | HEAO-2          |
| MRO / HiRISE |                | LRO / LROC NAC     | HERO            |
| OAO-2 / CEP  |                | SOHO/EIT           | FOXSI           |
| OAO-3 / PEP  |                | STEREO/SECCHI A    |                 |
| Planck       |                | TDRS-1             |                 |
| Proprietary  |                | TDRS-7             |                 |
| Spitzer      |                | TRACE              |                 |
| WIRE         |                |                    |                 |
| WISE         |                |                    |                 |
| WMAP         |                |                    |                 |

The MSFC ground OTA database contains information on 22 potential CERs (Table 3) for 26 telescopes from optical to radio (Table 4).

Table 3: Ground Telescope Database Cost, Programmatic and Engineering Parameters

| Primary Mirror Specific Information |             |  |  |  |  |  |  |
|-------------------------------------|-------------|--|--|--|--|--|--|
| PM Cost                             | \$ FY M     |  |  |  |  |  |  |
| PM Aperture Diameter                | meters      |  |  |  |  |  |  |
| PM Surface Figure Error             | rms nm      |  |  |  |  |  |  |
| PM Material                         |             |  |  |  |  |  |  |
| PM Focal Length                     | meters      |  |  |  |  |  |  |
| PM F/#                              |             |  |  |  |  |  |  |
| PM Number of Segments               | #           |  |  |  |  |  |  |
| PM Segment Size                     | meter       |  |  |  |  |  |  |
| PM Aspheric Departure               | micrometers |  |  |  |  |  |  |
| PM Mass                             | kg          |  |  |  |  |  |  |
| PM Lightweight Factor               | %           |  |  |  |  |  |  |

| Optical Telescope Assembly Information |             |  |  |  |  |  |  |
|--|-------------|--|--|--|--|--|--|
| OTA Cost                               | \$ FY M     |  |  |  |  |  |  |
| Diffraction Limit                      | micrometers |  |  |  |  |  |  |
| Transmitted WFE                        | nm rms      |  |  |  |  |  |  |
| Operating Temperature                  | K           |  |  |  |  |  |  |
| OTA Mass                               | kg          |  |  |  |  |  |  |
| Year of Development                    |             |  |  |  |  |  |  |
| Development Period                     | months      |  |  |  |  |  |  |
| Design Life                            | months      |  |  |  |  |  |  |
| On or Off-Axis                         |             |  |  |  |  |  |  |
| Number of Curved Optical Elements      |             |  |  |  |  |  |  |
| Optical Bench Material                 |             |  |  |  |  |  |  |

Table 4: Observatories in Ground Telescope Database

| In the Regression |
|-------------------|
| AEOS              |
| Commercial        |
| Commercial Radio  |
| DKIST             |
| Gemini 1          |
| Green Bank Radio  |
| HET               |
| JKT               |
| KECK 1            |
| KECK 1 & 2        |

| LBT                  |
|----------------------|
| Magellan 1           |
| MMT 6.5m replacement |
| SOAR                 |
| Starfire             |
| Subaru               |
| SubMM Array Dish     |
| UKIRT                |
| WHT                  |
| WIYN                 |
|                      |

| Not Included in Regression |
|----------------------------|
| ALOT                       |
| CHARA                      |
| DCT                        |
| IRTF                       |
| LAMP                       |
| VLA Dish                   |
|                            |
|                            |

Technical, programmatic and cost information for the database was collected from public reports, direct contact with project managers (via interviews and emails), and NASA archival sources:

- CADRe (Cost Analysis Data Requirements),
- NAFCOM (NASA/Air Force Cost Model) database.
- NICM (NASA Instrument Cost Model) database,
- NSCKN (NASA Safety Center Knowledge Now),
- RSIC (Redstone Scientific Information Center),
- REDSTAR (Resource Data Storage and Retrieval System) and
- SICM (Scientific Instrument Cost Model) database.

Of the 45 potential space CERs, there is sufficient data completeness to do pairwise cross-correlation (Table 5) for 15 potential variables. The purpose of this analysis is to identify CERs that are both correlated with cost and not correlated with each other. For example, cost is mostly highly correlated with 'size' parameters such as diameter, focal length and volume. This is logical, larger telescopes cost more than smaller telescopes. And, cost is highly correlated with launch date, which implies that more recent telescopes have been more expensive. But, launch date is also highly correlated with 'size' parameters, which implies that more recent telescopes have been larger. Thus, launch date is not an independent variable. The same applies to mass. Cost is correlated with mass, but mass is correlated with size, i.e. larger diameter telescopes have higher mass. It is, however, interesting to note that mass is correlated with design life. And, it is interesting to note the negative correlation between wavelength and operating temperature, i.e. longer wavelength systems operate at colder temperatures.

Table 5: Pair-Wise Cross-Correlation of 15 potential CERs for 18 imaging missions.

|                  | All Variable Pairwise Correlation Matrix for Space Imaging System Dataset (N=18) |            |        |          |           |       |       |       |             |                |                  |               | F              | Rev 12.05.2017 |               |  |
|------------------|--|------------|--------|----------|-----------|-------|-------|-------|-------------|----------------|------------------|---------------|----------------|----------------|---------------|--|
|                  | OTA\$  | Eff<br>Dia | Volume | PM<br>FL | Sys<br>FL | FOV   | WDLP  | Temp  | OTA<br>Mass | Design<br>Life | e^(YOD<br>-1960) | Dev<br>Period | Launch<br>Date | Orbit          | Point<br>Stab |  |
| OTA \$           | 1.00   | 0.85       | 0.94   | 0.98     | 0.98      | -0.22 | -0.15 | -0.07 | 0.78        | 0.59           | 0.00             | 0.59          | 0.93           | -0.18          | -0.27         |  |
| Eff Diameter     | 0.85   | 1.00       | 0.81   | 0.86     | 0.89      | 0.04  | 0.15  | 0.02  | 0.65        | 0.63           | 0.08             | 0.24          | 0.46           | -0.19          | -0.39         |  |
| Volume           | 0.94   | 0.81       | 1.00   | 0.94     | 0.92      | -0.11 | -0.06 | -0.22 | 0.54        | 0.36           | -0.03            | 0.53          | 1.00           | -0.11          | -0.31         |  |
| PM Focal Length  | 0.98   | 0.86       | 0.94   | 1.00     | 0.96      | -0.08 | -0.12 | -0.05 | 0.73        | 0.56           | 0.04             | 0.54          | 0.94           | -0.16          | -0.32         |  |
| Sys Focal Length | 0.98   | 0.89       | 0.92   | 0.96     | 1.00      | -0.22 | -0.12 | -0.13 | 0.78        | 0.63           | 0.29             | 0.63          | 0.90           | -0.11          | -0.30         |  |
| FOV              | -0.22  | 0.04       | -0.11  | -0.08    | -0.22     | 1.00  | 0.68  | -0.13 | -0.18       | -0.21          | -0.03            | 0.09          | -0.12          | -0.18          | -0.32         |  |
| WDLP             | -0.15  | 0.15       | -0.06  | -0.12    | -0.12     | 0.68  | 1.00  | -0.30 | -0.14       | -0.20          | -0.07            | 0.16          | -0.08          | -0.14          | -0.58         |  |
| Operate Temp     | -0.07  | 0.02       | -0.22  | -0.05    | -0.13     | -0.13 | -0.30 | 1.00  | 0.14        | 0.36           | 0.19             | -0.46         | -0.23          | 0.26           | 0.77          |  |
| OTA Mass         | 0.78   | 0.65       | 0.54   | 0.73     | 0.78      | -0.18 | -0.14 | 0.14  | 1.00        | 0.82           | 0.11             | 0.59          | 0.51           | -0.24          | -0.31         |  |
| Design Life      | 0.59   | 0.63       | 0.36   | 0.56     | 0.63      | -0.21 | -0.20 | 0.36  | 0.82        | 1.00           | 0.27             | 0.13          | 0.32           | 0.13           | -0.10         |  |
| e^(YOD-1960)     | 0.00   | 0.08       | -0.03  | 0.04     | 0.29      | -0.03 | -0.07 | 0.19  | 0.11        | 0.27           | 1.00             | -0.19         | 0.31           | -0.09          | -0.32         |  |
| Develop Period   | 0.59   | 0.24       | 0.53   | 0.54     | 0.63      | 0.09  | 0.16  | -0.46 | 0.59        | 0.13           | -0.19            | 1.00          | 0.50           | -0.19          | -0.75         |  |
| Launch Date      | 0.93   | 0.46       | 1.00   | 0.94     | 0.90      | -0.12 | -0.08 | -0.23 | 0.51        | 0.32           | 0.31             | 0.50          | 1.00           | -0.09          | -0.31         |  |
| Orbit            | -0.18  | -0.19      | -0.11  | -0.16    | -0.11     | -0.18 | -0.14 | 0.26  | -0.24       | 0.13           | -0.09            | -0.19         | -0.09          | 1.00           | 0.70          |  |
| Point Stability  | -0.27  | -0.39      | -0.31  | -0.32    | -0.30     | -0.32 | -0.58 | 0.77  | -0.31       | -0.10          | -0.32            | -0.75         | -0.31          | 0.70           | 1.00          |  |

The pairwise cross-correlation analysis identified eight potential CERs: aperture diameter, wavelength of diffraction limited performance (WDLP), operating temperature, year of development (YOD), primary mirror focal length, field of view, total mass and development period. Eighteen different combinations of these eight potential CERs were evaluated. And only four had a statistically significant (i.e. p < 10%) correlation with cost: effective aperture diameter, WDLP, operating temperature and YOD.

The MSFC telescope database has 100% completeness of these four CERs for 47 OTAs – 27 space and 20 ground. Table 5 gives the CER values for each of these OTAs. Cost data NASA proprietary.

Table 5: Space & Ground Telescope Database CERs

| SPACE TELESCOPES |                          |                      |                    |                        |                     |                        |                  | GRO                   | UND T                | ELESC  | COPES        |                   |          |
|------------------|--------------------------|----------------------|--------------------|------------------------|---------------------|------------------------|------------------|-----------------------|----------------------|--------|--------------|-------------------|----------|
| rev. 06.16.19    | Effective PM<br>Diameter | Diff. Lim. $\lambda$ | Operating<br>Temp. | Year of<br>Development | # of PM<br>Segments | PM Segment<br>Diameter | rev. 11.01.2018  | Effective<br>Diameter | Diffraction<br>Limit | Temp   | Year of Dev. | Total<br>Segments | Seg Size |
| Imaging          | (m)                      | (μ)                  | (K)                | (year)                 | #                   | (m)                    |                  | (m)                   | (µm)                 | K      | (year)       | #                 | (m)      |
| AFTA             | 2.40                     | 0.78                 | 284                | 1992                   | 1                   | 2.40                   | JKT              | 1.00                  | 1.00                 | 270.00 | 1977         | 1                 | 1        |
| COM_0.7          | 0.70                     | 0.50                 | 283                | 1996                   | 1                   | 0.70                   | Commercial       | 1.00                  | 0.50                 | 300.00 | 2013         | 1                 | 1        |
| COM_1.1          | 1.10                     | 0.65                 | 283                | 2007                   | 1                   | 1.10                   | Starfire         | 3.50                  | 0.53                 | 273.00 | 1989         | 1                 | 3.5      |
| Herschel         | 3.50                     | 80.00                | 80                 | 2001                   | 1                   | 6.50                   | WIYN             | 3.50                  | 0.42                 | 263.00 | 1988         | 1                 | 3.5      |
| HST              | 2.40                     | 0.50                 | 294                | 1977                   | 1                   | 2.40                   | AEOS             | 3.67                  | 0.85                 | 273.00 | 1991         | 1                 | 3.67     |
| IRAS             | 0.57                     | 8.00                 | 4                  | 1977                   | 1                   | 0.57                   | UKIRT            | 3.80                  | 2.20                 | 273.00 | 1974         | 1                 | 3.8      |
| JWST             | 6.20                     | 2.00                 | 50                 | 2006                   | 18                  | 1.40                   | SOAR             | 4.20                  | 1.00                 | 263.00 | 1997         | 1                 | 4.2      |
| Kepler           | 1.40                     | 1.00                 | 213                | 2001                   | 1                   | 1.40                   | WHT              | 4.20                  | 6.10                 | 270.00 | 1981         | 1                 | 4.2      |
| MO / MOC         | 0.35                     | 0.53                 | 283                | 1986                   | 1                   | 0.35                   | DKIST            | 4.20                  | 0.90                 | 300.00 | 2011         | 1                 | 4.2      |
| MRO / HiRISE     | 0.50                     | 0.40                 | 293                | 2001                   | 1                   | 0.50                   | MMT Replace      | 6.50                  | 1.60                 | 262.00 | 1992         | 1                 | 6.5      |
| OAO-2 / CEP      | 0.61                     | 1.50                 | 300                | 1962                   | 4                   | 0.31                   | Magellan 1       | 6.50                  | 1.00                 | 280.00 | 1994         | 1                 | 6.5      |
| OAO-3 / PEP      | 0.80                     | 2.40                 | 288.5              | 1963                   | 1                   | 0.80                   | Gemini 1         | 8.10                  | 0.80                 | 270.00 | 1994         | 1                 | 8.1      |
| Planck           | 1.70                     | 300.00               | 40                 | 2001                   | 1                   | 1.70                   | Subaru           | 8.30                  | 0.60                 | 273.00 | 1988         | 1                 | 8.3      |
| Proprietary      |                          |                      |                    |                        |                     |                        | KECK 1           | 10.00                 | 1.00                 | 273.00 | 1986         | 36                | 1.8      |
| Spitzer          | 0.85                     | 6.50                 | 5.5                | 1995                   | 1                   | 0.85                   | LBT              | 11.88                 | 0.65                 | 273.00 | 1997         | 2                 | 8.4      |
| WIRE             | 0.30                     | 24.00                | 12                 | 1995                   | 1                   | 0.30                   | KECK-I&II        | 14.14                 | 1.00                 | 273.00 | 1986         | 72                | 1.8      |
| WISE             | 0.40                     | 2.75                 | 17                 | 2002                   | 1                   | 0.40                   | HET              | 9.20                  | 20.00                | 264.00 | 1994         | 91                | 1        |
| WMAP             | 2.10                     | 1300.00              | 60                 | 1996                   | 2                   | 1.50                   | Commercial Radio | 5.00                  | 210000.00            | 300.00 | 2012         | 1                 | 5        |
| Non-Imaging      |                          |                      |                    |                        |                     |                        | SubMM Array Dish | 6.00                  | 300.00               | 300.00 | 1998         | 72                | 1        |
| ACTS             | 3.97                     | 1950.00              | 263                | 1984                   | 2                   | 2.80                   | Green Bank Radio | 100.00                | 6500.00              | 300.00 | 1991         | 2004              | 3        |
| CALIPSO          | 1.00                     | 6.60                 | 283                | 2000                   | 1                   | 1.00                   |                  |                       |                      |        |              |                   |          |
| Cloudsat         | 1.85                     | 1300.00              | 250                | 2000                   | 1                   | 1.85                   |                  |                       |                      |        |              |                   |          |
| GALEX            | 0.50                     | 8.00                 | 273                | 1998                   | 1                   | 0.50                   |                  |                       |                      |        |              |                   |          |
| ICESat           | 1.00                     | 8.00                 | 283                | 1998                   | 1                   | 1.00                   |                  |                       |                      |        |              |                   |          |
| IUE              | 0.45                     | 3.50                 | 273                | 1973                   | 1                   | 0.45                   |                  |                       |                      |        |              |                   |          |
| MO / MOLA        | 0.50                     | 15.00                | 283                | 1986                   | 1                   | 0.50                   |                  |                       |                      |        |              |                   |          |
| OAO-B / GEP      | 0.97                     | 5.00                 | 289                | 1964                   | 1                   | 0.97                   |                  |                       |                      |        |              |                   |          |
| SWAS             | 0.68                     | 286.00               | 170                | 1993                   | 1                   | 0.68                   |                  |                       |                      |        |              |                   |          |

To insure a meaningful statistical regression, much effort was expended to compile a database with a wide range of data diversity. To be specific, we added radio and sub-millimeter telescopes for wavelength diversity and located cost and technical information for the Orbiting Astronomical

Observatory #2 Celeste Experiment Package (OAO-2/CEP) and the OAO-3 Princeton Experiment Package (OAO-3/PEP). And we added CALIPSO and DKIST.

For the 20 Ground Telescopes:

- Diameter ranges from 1 meter to 100 meters
- WDLP ranges from 500 nm to 21 centimeters
- YOD ranges from 1979 to Present
- 14 Monolithic and 6 Segmented

# 27 Space Telescopes

- Diameter ranges from 30 cm to 6.35 meter
- WDLP ranges from 400 nm to 2 mm
- Operating Temperature ranges from 4 to 300K
- YOD ranges from 1962 to Present
- 23 Monolithic and 4 Segmented
- 18 Imaging and 9 Non-Imaging

Please note that effective Diameter may not be the same as the telescope's aperture stop diameter. For example, the Kepler PM is larger than the stop defined by the Schmidt corrector. And, WMAP, LBT and Keck I&II all have two telescopes. Their effective PM diameter is the size of a circular aperture with the same total collecting area.

Also, in compiling the database, it was discovered that we were using different YOD definitions for space and ground. When we started the cost modeling effort in 1999, we defined YOD to be 'first light' for ground telescopes. We considered defining YOD to be 'launch date' for space telescopes but used start of Phase C instead – because launch dates can be delayed for no fault of the missions, i.e. Hubble's launch date was delayed due to the Challenger accident. But, when we combined the two databases, we discovered that there was approximately a 6 year effective difference between these two definitions and that difference resulted in a difference in the Ground-to-Space scale factor. Thus, we had to update the ground telescope database with the dates for when contracts were issued for each telescope.

## **Cost Model Regression**

The MSFC OTA database was regressed against the 4 CERs plus a Space/Ground multiplier factor to yield a 5-parameter cost model that explains 92% (Adjusted R2) of the cost variation in a database of 47 total ground and space telescope assemblies.

|          |                                  | (S/G)     | ) (  | <b>(1.7</b> ) | (-0.5)      | (-0.25 | 5) (-(  | 0.028) (Y-1960) |
|----------|----------------------------------|-----------|------|---------------|-------------|--------|---------|-----------------|
| OTA\$ (F | $\mathbf{Y17}) = \$20\mathbf{M}$ | x 30      | x D  | X             | $\lambda$ x | T      | x e     |                 |
|          | Parameter                        | Intercept | S/G  | D             | λ           | Т      | YOD     |                 |
|          | Model Value                      | 20        | 30   | 1.7           | -0.5        | -0.25  | -0.028  |                 |
|          | Actual Value                     | 20.7      | 28.1 | 1.695         | -0.466      | -0.256 | -0.0282 |                 |
|          | SE                               | 1.6       | 6    | 0.09          | 0.03        | 0.07   | 0.006   |                 |

8E-21

6E-21

0.001

3E-05

1E-18

where: (S/G) = 1 for Space OTAs

p-value

= 0 for Ground OTAs

D = Effective Telescope Aperture Diameter

2E-07

λ = Wavelength of Diffraction Limited Performance

T = Operating Temperature

YOD = Year of Development, i.e. start of Phase C/D or Issue of Contracts.

Please note that regressions with different combinations of variables and dataset (i.e. ground or space only, or monolithic or segmented only) yields different exponent values for the model parameters. It is for this reason, that one should never blindly use a cost model without understanding the database from

which it is constructed. For example, if one regresses only on effective diameter for the entire MSFC database, its exponent is close to 1.0. The reason is because the largest apertures in the database are radio and sub-mm wave telescope, which cost less. Adding WDLP to the regression increases the aperture exponent into the 1.6 range. Alternatively, you can get the same aperture exponent by restricting the regression to only optical wavelength telescopes. Temperature only has an impact for space telescopes – because ground telescopes operate close to ambient. Finally, there is an interesting relationship between aperture diameter and YOD. Adding YOD increases the aperture diameter exponent. The reason is because more recent telescope tend to be larger and – because of technology advance – they tend to have a lower cost per area. Thus, without the YOD term, one gets a 'flatter' trend line for cost versus aperture diameter. This effect is particularly noticeable if you create a database subset of the first occurrence of each aperture 'size' for ground telescopes; without YOD the diameter exponent is 1.3 and with YOD the diameter exponent is 1.5.

## **Residual Analysis**

The 'goodness' of the model was evaluated via residual and outlier analysis. Each column in Figure 1 shows cost versus CER (diameter, wavelength, temperature and YOD). The top row plots 'raw' database cost vs each CER. At best one can say that there is a slight diameter trend line and that cost per diameter shows a slight space vs ground clustering. Row two plots the raw data normalized by diameter. Once the data is normalized for diameter, there is an obvious wavelength dependency. And, there are obviously two trend lines in wavelength, temperature and YOD (for ground and space). Row three adds wavelength normalization. At this point, a clear diameter trend emerges and the ground vs space clusters are solidified. Normalizing by temperature and YOD in row four tightens the groupings. And, scaling by Space/Ground in row five pulls the two datasets into a single model.



performance of 3.19-mm, the telescope was actually built with a WDLP of 1.3-mm. Similarly, we found a better citation for Planck and changed its WDLP from 700 to 300 micrometers.

### **Segmented Aperture Cost Model**

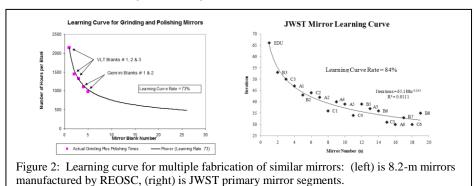
Given that telescope cost is primarily driven by aperture diameter. And, that there is a practical limit to how large of a monolithic mirror one can make. The question of mirror segmentation must be addressed.

Historically, segmentation has always been the solution for when technology did not allow a monolithic mirror. But as soon as technology permitted, segmented mirrors were replaced with monoliths. The original 'Large Space Telescope' (i.e. Hubble) was a segmented mirror. Then NASA funded the development of lightweight high-temperature-fused ULE mirrors. The Multi-Mirrored Telescope has been replaced by a 6.5m monolith. And, 10-m class segmented telescopes such as Keck gave way to 8-m class monoliths such as VLT and Subaru. Finally, JWST's 6.2-m aperture was segmented because it had to fit inside a 4.5-m launch fairing. If the Space Launch System (SLS) had existed at the time that JWST was being developed, it may well have had a monolithic mirror.

The MSFC database as a total of 10 segmented telescope (6 ground and 4 space). Because the segmented telescope database is small, it is difficult to perform meaningful regressions. Potentially a 'trick' is to consider the 36 monolithic telescopes as 'one' segment apertures. Replacing the Effective Diameter (D) parameter with a segmentation parameter (Nseg x Dseg, where Nseg = the number of segments in the aperture and Dseg = the segment circumscribed diameter) and regressing on the 47 telescope MSFC database yields a potential 6-variable ground and space segmented telescope cost model.

|             |              | (S/G)     | ((    | 0.8)  | (1.7) | (-0.       | 5)     | (-0.25) | (-0.028) (Y-1960) |
|-------------|--------------|-----------|-------|-------|-------|------------|--------|---------|-------------------|
| OTA\$ (FY17 | ) = \$20M x  | 30 x      | Nseg  | x D   | seg   | <b>x</b> λ | x T    | X       | e                 |
|             | Parameter    | Intercept | S/G   | Nseg  | Dseg  | λ          | T      | YOD     |                   |
|             | Model Value  | 20        | 30    | 0.8   | 1.7   | -0.5       | -0.25  | -0.028  |                   |
|             | Actual Value | 22.0      | 25.5  | 0.78  | 1.63  | -0.47      | -0.248 | -0.0287 |                   |
|             | SE           | 1.7       | 1.3   | 0.06  | 0.11  | 0.03       | 0.08   | 0.003   |                   |
|             | p-value      | 1E-06     | 1E-16 | 2E-16 | 5E-17 | 5E-19      | 0.004  | 1E-04   |                   |

Analysis with different database combinations indicates a correlation between the exponents for Nseg and Dseg. Their solutions range from Nseg<sup>(0.78)</sup> Dseg<sup>(1.61)</sup> to Nseg<sup>(0.84)</sup> Dseg<sup>(1.73)</sup>. For the segmented aperture cost model, we choose Nseg<sup>(0.8)</sup> Dseg<sup>(1.7)</sup>. The Dseg exponent should be the same value as for D in the monolithic model. The cost to make a single mirror should be the same in both model. And the 0.8 exponent for Nseg is consistent with empirical data from the manufacture of 8-m monolithic mirrors by REOSC and 1.4-m JWST mirror segments (Figure 2).



Finally, just as regression analysis identified the S/G scale factor, we investigated if there was a scale factor between monolithic and segmented mirrors. The regression 'indicates' that segmented aperture telescopes are approximately 15% more expensive than monolithic telescopes. BUT, the regression result is not significant. It has a p-value = 30%. So the segmented cost model assumes no difference.

## **Cost Estimating Examples**

OTA cost can be estimated via two methods: using the model directly (Figure 3) or using the model to compare relative cost with other OTAs (i.e. Hubble & JWST) (Figure 4). Please note - remembering the discussion that segmented aperture telescopes 'may be' 15% more expensive than monolithic, but such a scaling factor was not included in the model because its regression was statistically uncertain - that it is possible that the model estimates for segmented aperture telescopes are low. And, maybe the estimates based on relative ratios of JWST are more correct.

|                                |        | Equation Method |           |          |          |  |  |
|--------------------------------|--------|-----------------|-----------|----------|----------|--|--|
| Aperture                       |        | 4-m             | 3.6-m Seg | 8-m      | 8-m Seg  |  |  |
| Architecture                   |        | off-axis        | on-axis   | off-axis | off-axis |  |  |
| Starting Space Cost [FY17 \$M] | \$ 600 |                 |           |          |          |  |  |
| Number of Segments             | 0.8    | 1               | 6         | 1        | 37       |  |  |
| Diameter [meter]               | 1.7    | 4               | 1.5       | 8        | 1.5      |  |  |
| WDLP [micrometer]              | -0.5   | 0.4             | 0.4       | 0.4      | 0.4      |  |  |
| Temperature [K]                | -0.25  | 270             | 270       | 270      | 270      |  |  |
| exp(YOD)                       | -0.028 | 2025            | 2025      | 2025     | 2025     |  |  |
| 50% Predicted Cost [FY17 \$M]  |        | \$ 400          | \$ 317    | \$ 1,301 | \$ 1,358 |  |  |
| 85% Predicted Cost [FY17 \$M]  |        | \$ 580          | \$ 459    | \$ 1,886 | \$ 1,969 |  |  |

In Figure 4, the ratio column indicates whether the telescope whose cost is being estimated is more or less expensive than the basis of estimate telescope. For example. A 4-m monolithic aperture telescope is 2.4X more expensive than HST – based on aperture – but half as expensive as JWST. A UVO telescope will be approx. 2X more expensive than JWST but an ambient temperature telescope will be approximately half as expensive, so those parameters tend to cancel. Finally regarding YOD, technology has advanced since HST was started in 1977 that an identical telescope should cost only 25% of the original in 2025 and if a second JWST telescope were made in 2025, it should cost only 60%.

|                               |        |       | Relative Cost Method |       |         |          |       |         |          |       |
|-------------------------------|--------|-------|----------------------|-------|---------|----------|-------|---------|----------|-------|
| Aperture                      |        |       | 4-m Mono             |       |         | 4-m Mono |       | JWST    | 8-m Seg  |       |
| Architecture                  | :      | HST   | off-axis             | Ratio | JWST    | off-axis | Ratio |         | off-axis | Ratio |
| Total Cost [FY17 \$M]         |        | \$530 |                      |       | \$1,380 |          |       | \$1,380 |          |       |
| Number of Segments            | 0.8    | 1     | 1                    | 1.00  | 1       | 1        | 1.00  | 18      | 37       | 1.78  |
| Diameter [meter]              | 1.7    | 2.4   | 4                    | 2.38  | 6.35    | 4        | 0.46  | 1.4     | 1.4      | 1.00  |
| WDLP [micrometer]             | -0.5   | 0.5   | 0.4                  | 1.12  | 2       | 0.4      | 2.24  | 2       | 0.4      | 2.24  |
| Temperature [K]               | -0.25  | 294   | 270                  | 1.02  | 50      | 270      | 0.66  | 50      | 270      | 0.66  |
| exp(YOD)                      | -0.028 | 1977  | 2025                 | 0.26  | 2006    | 2025     | 0.59  | 2006    | 2025     | 0.59  |
| 50% Predicted Cost [FY17 \$M] |        |       | \$376                | 0.71  |         | \$ 542   | 0.39  |         | \$ 2,116 | 1.53  |
| 85% Predicted Cost [FY17 \$M] |        |       | \$546                |       |         | \$ 786   |       |         | \$ 3,069 |       |

Figure 4: Examples predicted telescope cost estimates using model to ratio relative to HST and JWST.

#### **Mass Cost Models**

While many cost models use mass as a CER, our research finds that mass is not a good CER. The problem with mass is that it is not an independent variable. It is correlated with aperture. The best example of the inappropriateness of mass is the fact that at 3180 kg HST's OTA is 1.5X more massive than JWST's 2170 kg OTA. Thus, a mass-based cost model would estimate JWST to be 66% the cost of HST when in reality JWST was approximately 2.6X more expensive than HST.

While the following is supposition on the part of the author and cannot be tested, one reason for the difference between HST and JWST cost relative to their mass may be launch vehicle mass capacity and design margin. Because the space shuttle had a launch mass of 16,000 kg to LEO while the Ariane-5 can only launch 6600 kg to SE-L2, JWST had to be designed to a lower mass than HST. The advantage of having extra mass available is that it enables stiffer, more robust and stable optical components and structure – which are easier to manufacture with less risk and lower total cost.

# **Subsystem Cost**

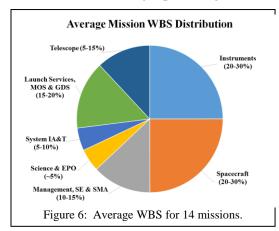
A final topic is the connection between telescope cost and total mission cost. Many believe that telescope cost drives mission cost. To test this belief, cost data was collected in the standard NASA work breakdown structure (WBS) (Figure 5) from Cost Analysis Data Requirements (CADRe) reports for 14 missions:

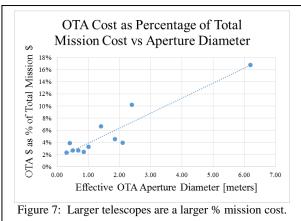
- CALIPSO
- CLOUDSAT
- GALEX
- ICESAT
- JWST
- Kepler
- LANDSAT-7
- Spitzer

- STEREO
- SWAS
- TRACE
- WIRE
- WISE
- WMAP



As summarized in Figure 6, the telescope is only about 5 to 15% of the total mission cost. And this percentage is highly dependent on the telescope's aperture diameter (Figure 7). For the 14 missions for which we have complete WBS cost data, the science instruments and spacecraft are each a larger percentage of the total mission cost. Furthermore, management, systems engineering and safety and mission assurance are a larger percentage of total mission cost than the telescope.





## Conclusion

After 20 years assembling and vetting a database for ground and space telescopes containing cost, programmatic and technical data, we have sufficient quantity of telescopes with sufficient data diversity to develop a multivariable parametric cost model that explains 92% of the cost variation for the ground and space telescopes in the data base. This model can be used to provide a basis of estimate for a given telescope concept, compare between major architectural cost drivers, and justify technology development investment. This white paper summarized the MSFC database and presented the cost model it supports. The white paper also – for the first time – presents a potential extension to the model for segmented aperture telescopes. Examples of how the model can be used to estimate telescope concept costs via either direct application or relative comparison to historical exemplars are presented. The White Paper wishes to reiterate that mass is not a good cost estimating relationship for telescopes. Instead, telescope cost is driven by its effective diameter, wavelength of diffraction limited performance, operating temperature and year of development. Finally, telescope cost is not the driving cost for mission in our database. Instead, science instruments and spacecraft are a larger cost driver. Furthermore, management is typically a larger cost driver than the telescope.

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